

Instrumentation and analysis strategies to search for particle precipitation from space

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Talk overview

Instrumentation and analysis strategies to search for particle precipitation from space

VOLTA Electric Field Detector for the CSES satellite

SAMPEX-PET mission and database

Main characteristics of the SAMPEX/PET mission:

Orbit altitude:	520÷670 km
Orbit inclination:	82°
PET Pointing modes:	ORR; MORR; 1 RPM (see text)

PET channel Level-2 data used for this study

Particles	Energy (MeV)	Geometric factor $(m^2 sr)$	Channel
Protons	28÷60	1.5	PHI
Protons	19÷28	1.65	PLE
Electrons	2÷6	1.65	ELO
Electrons	4÷15	1.5	EHI
Electrons	4÷30	-	EWG
Protons	> 60	0.4	RNG
Electrons	> 15	0.7	
Protons	> 85	N 25	PEN
Electrons	> 30	U.2J	ГĽІЧ

SAMPEX Pointing Modes

SAMPEX/PET has operated with three different pointing programs:

•ORR	(original Orbit Rate Rotation)
•MORR	(Modified Orbit Rate Rotation)
•1 RPM	(1 Rotation Per Minute)

 \checkmark During the <u>ORR pointing mode the PET yaw axis is substantially radial to the Earth</u>. So, PET may detect particles with pitch angle in a wide range and, in particular, also in the loss cone (precipitating particles) or near to it.

✓ On the contrary, in the MORR mode the detector yaw axis is fundamentally perpendicular to the geomagnetic field lines, since it was implemented mainly to study particles with pitch angle near 90° (trapped particles). Measurements for α_{PET} values far from 90° are performed in periods during which PET yaw axis is parallel to the geomagnetic field (B), when B > 0.3 G, and perpendicular to it, when B < 0.3 G.

✓ Finally, in the 1 RPM mode the α_{PET} distribution is flat since <u>the PET yaw axis, rotating continuously at</u> <u>1 RPM</u>, allows the particle detection at **any pitch angle value**.

Particle Burst definition

• Observations:

PBs are particle fluxes characterized by an **anomalously sharp increase and time duration of a few ten seconds**. In a sequence of CRs (each one having a duration of 30 seconds) we observe a PB as an increase in the number of detected particles.

• Duration:

The duration of this time series of data is relatively short, since **it only consists of several CRs**. The 30s temporal resolution of Level-2 PET-data does not allow characteristic PB events to be individually observed. Then, only an anomalous deviation from the statistical distribution of the 30s integrated CRs can be identified.

• Parameters:

These PB events are defined beginning from CRs detected at known L-shell and α_{PET} values.

From the Counting Rates to the Particle Bursts

Construction of the *"Background matrix"*

- L-shell and B values for each CR detection position are calculated by the IGRF.
- Analysed L-shell interval: $1.0 \le L$ -shell ≤ 1.8
- **Background matrix:** $0^{\circ} \le \alpha_{\text{PET}} \le 180^{\circ}$ (step = 15°) & $1 \le \text{L-shell} \le 1.8$ (8 steps of 0.1)

CRs Poissonian selection creteria \rightarrow **PBs definition**

✓ For each cell of the {L, α_{PET} } matrix, the CR distribution is the reference

background for estimating PBs.

CRs have a Poissonian distribution for each $\{L, \alpha_{PET}\}$ matrix cell.

✓ We define as PBs those counting rates which:

 assume <u>values greater than those of the relative</u> {L,α_{PET}} 	<u>background matrix</u>			
&				
probability to be a Poisson fluctuation is less than < 0.01.				

Particle Bursts selection criteria

Constraints:

➤ The lower boundary of the inner radiation belt (and in particular the SAA region) has been determined yearly by the IGRF

Exclusion of stably trapped particles \Rightarrow PBs are constituted by particle precipitating fluxes !!

 \prod

The **SAA region has been excluded** from this study.

➢ In order to reduce ionospheric and magnetospheric contamination from non-seismic sources, we use only data collected:

- during periods characterized by a geomagnetic index value $A_P < 20$
- in absence of ionospheric disturbances (SID=0).

Particle Burst Catalogue

- Particle Burst definition:
 - \checkmark CRs = particle flux measurements every **30 seconds**
 - ✓ CR background matrix {L, α_{IDP} }:
 - \checkmark 1 \leq L-shell \leq 1.8 (step=0.1)
 - $\checkmark \qquad 60^{\circ} \le \alpha_{IDP} \le 120^{\circ} \text{ (step = 15^{\circ})}$

✓ Poissonian cut p < 0.01

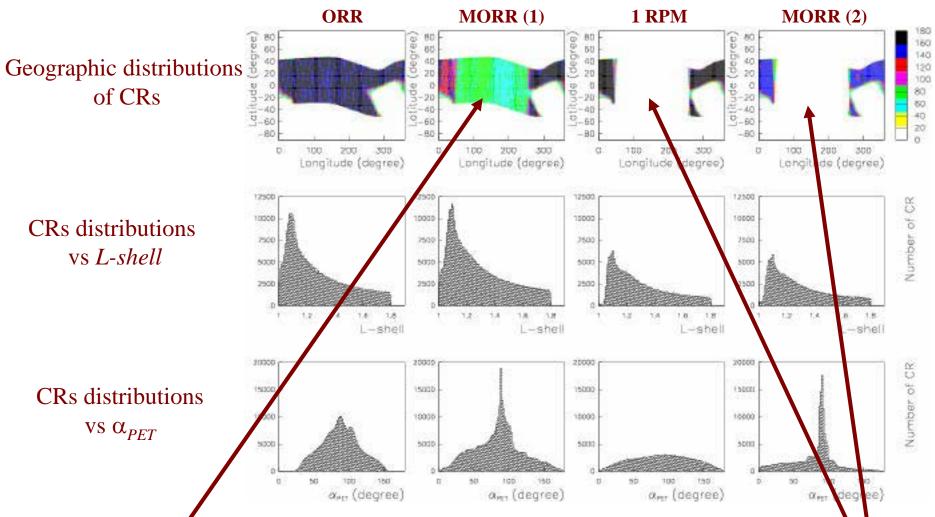
 \checkmark No measurements in the SAA reagion

 $\checkmark A_{\rm P} < 20$ & SID = 0

• **PBEHI**& **PBELO** catalogues for the SAMPEX pointing mode periods:

Pointing mode	Period (MM/DD/YY)	# "good days"	#CR	#CR per day	#PBEHI	#PBELO
ORR	01/24/93÷05/26/94	247	211603	857	1734	1259
MORR (1)	05/27/94÷05/07/96	691	373976	541	3957	1848
1 RPM	05/08/96÷05/06/98	677	195187	288	1253	485
MORR (2)	05/07/98÷12/16/99	543	179457	330	1628	698

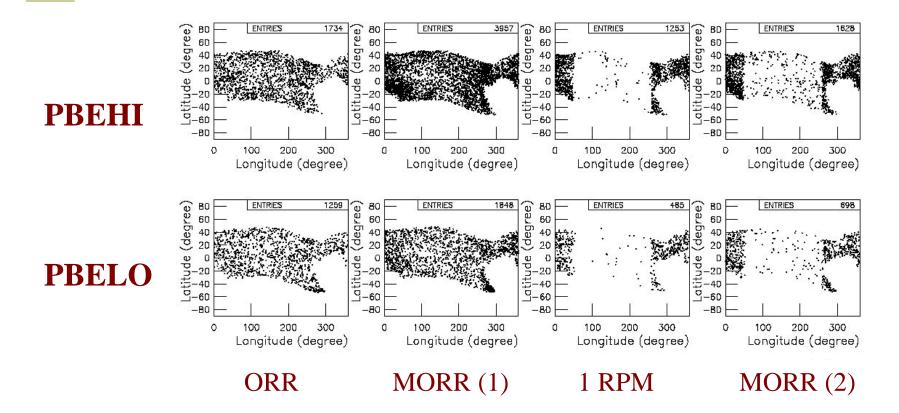
Reported daily data have been collected with **optimal device working conditions** (here referred to as "**good days**") and detected with good quality flag according with the SAMPEX weekly instrument status report.



SAMPEX measurements for different pointing modes

- During the MORR (1) mode there is a non-homogeneous CRs geographical distribution due to scanty PET measurements in the longitude interval [50°-270°].
- PET measurements are further reduced in this longitudinal interval during the 1RPM and MORR(2) pointing programs.
- The 4 panels in the bottom confirm that the α_{PET} distribution is pointing program dependent.

Geographic PBs distributions



Geographic distributions of **PBEHI** (top) & **PBELO** (bottom) data in the 1.0 < L < 1.8 interval for the 4 pointing modes.

Earthquakes catalogue

Earthquake selection (USGS catalogue):

- M ≥ 5.0
- hypocentral depth ≤ 100 km

Seismic EME & particle interaction altitude:

 $100 \ km \leq \ Z_{\rm EME} \ \leq 1200 \ km$

 $\begin{aligned} & \hspace{0.5cm} \succeq \underline{ Earthquake, } L_{\underline{EME}} = \underline{ L- shell of the interaction zone between} \\ & \underline{ trapped particles \& preseismic EME} \end{aligned}$

$$\mathbf{L}_{\mathbf{EME}} = \mathbf{L} | \mathbf{Z}_{\mathbf{EME}} = 400 \, \mathbf{km}$$

Selection criteria for the EQs and PBs populations to be correlated

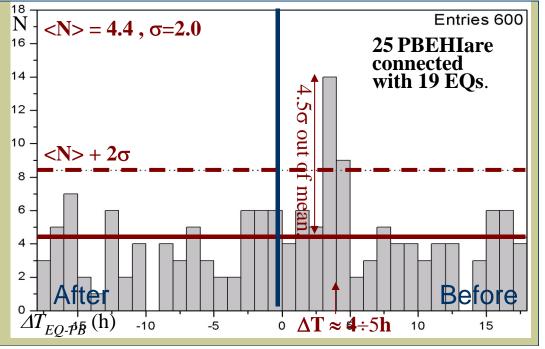
EQ event selection	tion PB event definition	
M≥5.0	CRs = particle flux measurements integrated over 30:	
Hypocentral depth h < 100km	CR daily background matrix $\{L, \alpha_{PET}\}$:	
	$1.0 \le L_{\rm PB} \le 1.8 \ (L\text{-step} = 0.1)$	
$L_{EME} \sim L_{z=400 \mathrm{km}}$	$0^{\circ} \le \alpha_{PET} \le 180^{\circ} (\alpha_{PET} \text{-step} = 15^{\circ})$	
	Poissonian cut: p < 0.01	
	No measurements in the SAA region	
	$A_p < 20$	
	SID = 0	
EQ-PB correlation		
$\Delta L = L_{EME} - L_{PB} < 0.1$		
$\Delta T_{MAX} = (T_{EQ} - T_{PB})_{MAX} = \pm 3 \text{ days}$		

SAMPEX: ΔT_{EO-PB} correlations between PBs and Eqs 211603 CR, 1734 PBEHI, 2074 EQs-PBs, Ν ORR mode 35 $|\Delta T_{EO-PB}|_{MAX} = \pm 3$ days Only this peak (31 events) • The existence of a $\Delta T > 0$ peak 30 is 4σ out of mean Bin step = 1h. suggests that **PB events statistically** Histogram window = ± 18 h. 25 precede EQs occurrence. $\leq N \geq + 2\sigma$ 20 • ΔT maximum positive peak has $<N> = 15.5, \sigma = 3.5$ been observed: 15 - only when considering **PBEHI** 10 data obtained during the **ORR** mode 5 -Before **\tte** - Not maximum positive peak is found 0 -10 -5 Ó 5 10 15 for PBELO data. $\Delta T \approx 4h$ ΔT_{EO-PB} (h)

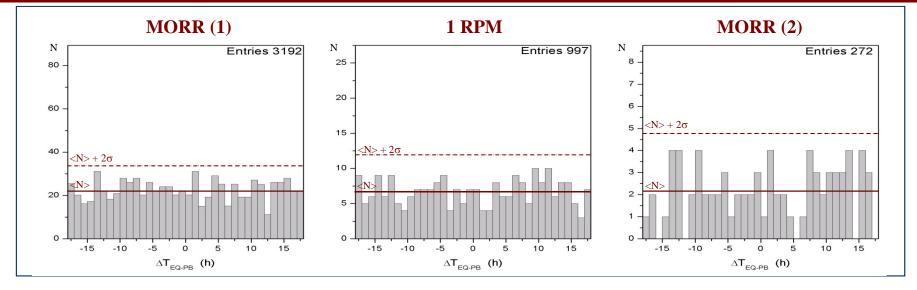
Improvement of the ΔT_{EQ-PB} **correlations with** αPET **cut**

An increase of the peak ΔT_{EQ-PB} statistical significance in the ORR histogram is obtained by excluding data of charged particles with $\alpha_{PET} \approx 90^{\circ}$ (trapped particles). Histogram is obtained by

<u>excluding</u> α_{PET} range: 70° < α_{PET} <110°

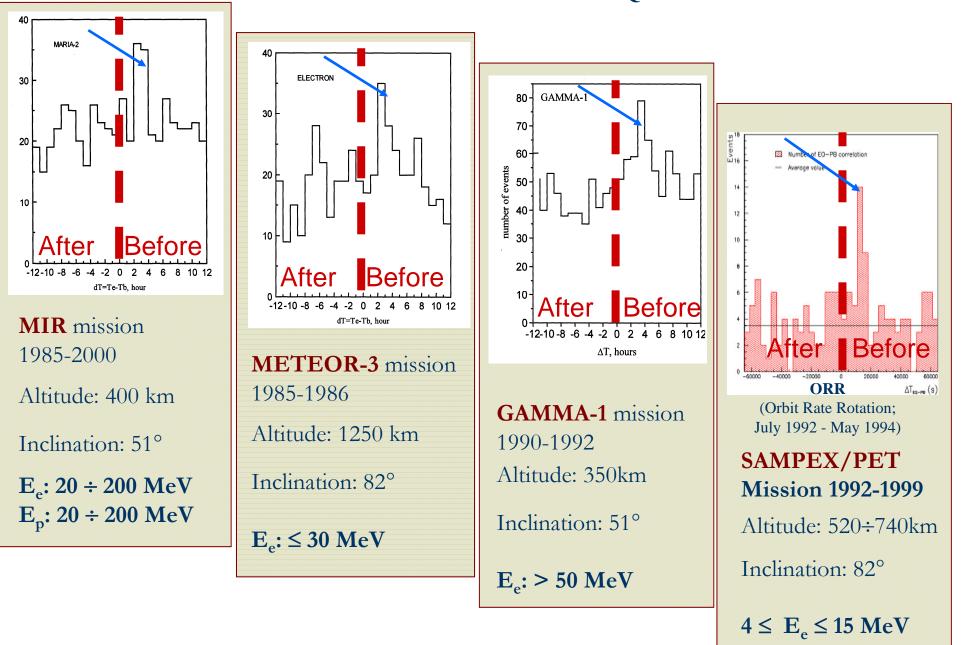


Dependence of the ΔT_{EQ-PB} correlation on: the particles energy & the PET yaw axis orientation



- ✓ No correlation is obtained (that is, no relevant peak is observed) with PBEHI data collected in the other MORR(1), 1RPM, and MORR(2) pointing periods
- ✓ **No correlation** is obtained in the 4 pointing mode periods with **PBELO** data.
 - The good correlation obtained only with ORR PBEHI suggest a <u>dependence</u> <u>from the particle energy</u> and confirms results reported in literature.
 - Our study results have demonstrated also a <u>dependence of the ΔT_{EQ-PB} </u> correlation on the **PET yaw axis orientation**

Correlations between EQ & ps: ΔT_{EQ-PB} distributions



Discussion & conclusions

• PBs of precipitating high-energy Van Allen electrons appear to precede statistically by some hours the occurrence of moderate and strong EQs.

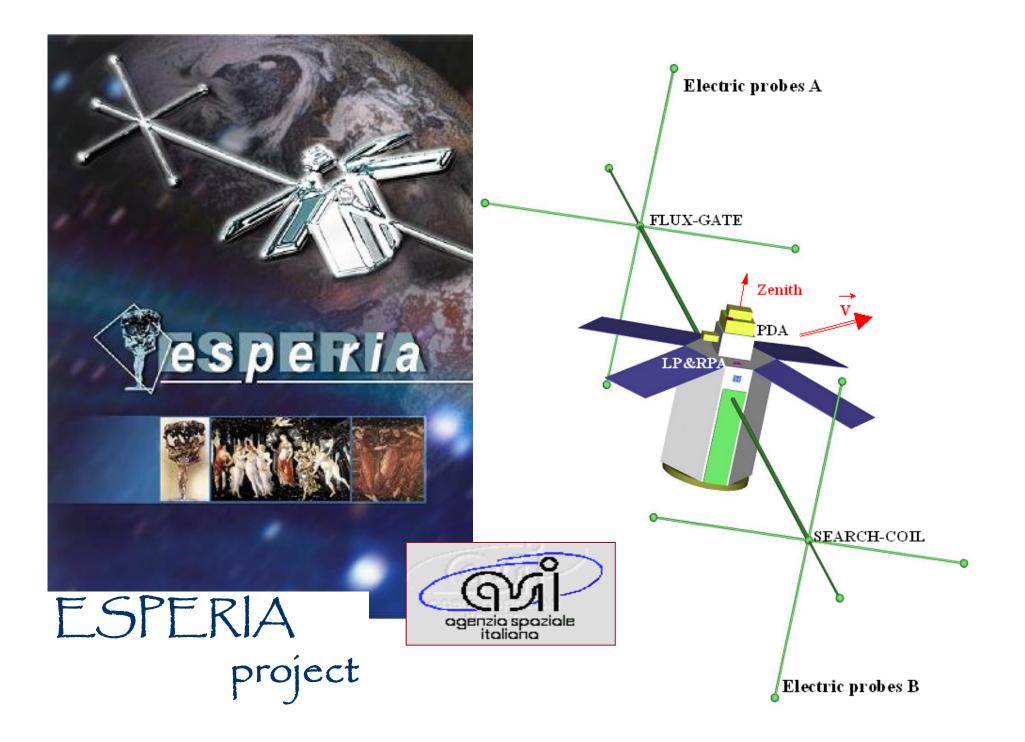
• Several constrains and cuts have been applied to data in order to exclude correlation with non-seismic phenomena.

• No correlation was found between PBs and other nonseismic sources.

• The results we have obtained:

✓ give indication for a **deeper investigation of the physical mechanisms** under study

✓ point out the importance of satellite pointing modes, pitch angle reconstruction, high energy PDA, *detection of both electrons and protons* for longitude reconstruction, etc.

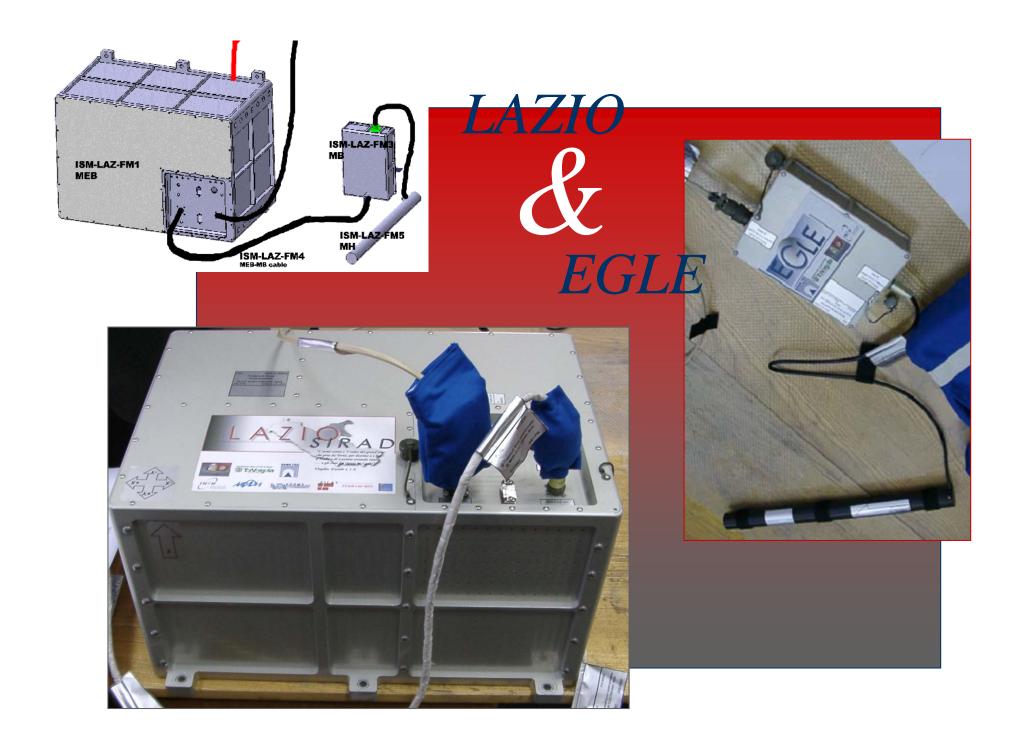




"L'armi canto e 'l valor del grand'eroe che pria da Troia, per destino a i liti d'Italia e di Lavinio errando venne; ... e gli suoi dèi ripose in Lazio"

Virgilio, Eneide I, 1-9.





LAZIO-EGLE equipment

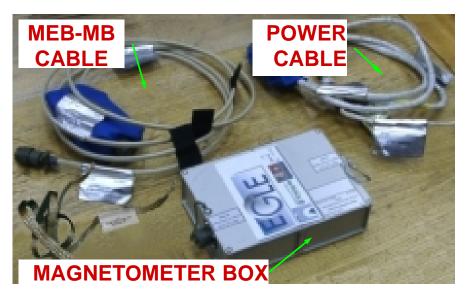


Main Electronic Box

MAGNETOMETER HEAD







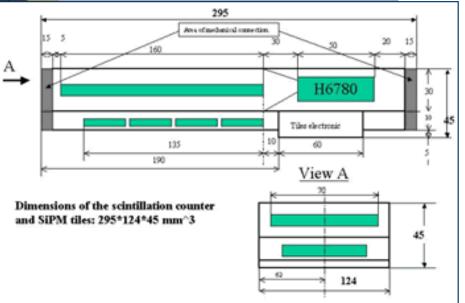
L AZIO Particle Detector Analyser

LAZIO particle detector

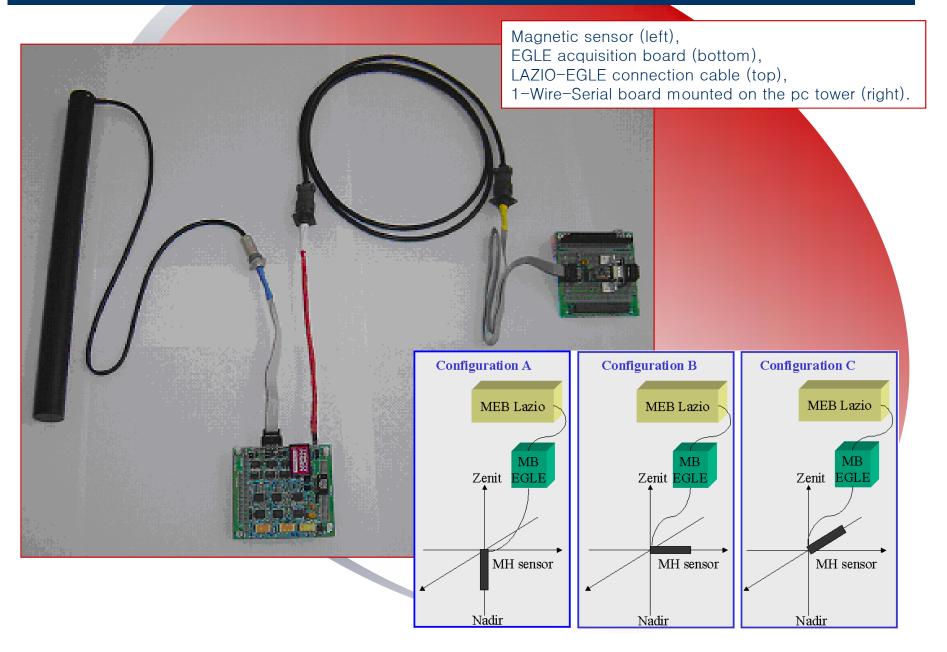




LAZIO scintillators layout



EGLE includes a *single axis search-coil magnetometer MH*



EGLE magnetometer

Basic technical specifications of the EG	LE probe MH		
Frequency b and of receiver signals	0.5 ÷ 50000 1	Hz	
Shape of transfer function	linear — flat		
Type of output	Symmetrical		
T ransformation factor at both output terminals:			
at linear part (0.5 – 5 Hz)	f*4 mV/(nT*	*Hz)	
at flat part (5 – 50000 Hz)	20 mV/nT	S	
Transformation factor error:		mV/nT	
at flat part of band pass without edges	≤±0.25 dB		
at flat part band pass edges	≤3 dB	10	
Magnetic noise level, pT *Hz ^{-1/2} :			-3 dB -3 dB
at 5 Hz	101		
	≤0.4 ≤0.02	1,0	
 at 100 Hz 	≤ 0.02		40
at 5 kHz	≤ 0.004		
at 50 kHz	≤0.02	0,1 l	
Nominal output load	≤ 200 pF	0.	.1 1 10 100 1000 10000 f, Hz
Nominai ou p ut ioau	\geq 50 k Ω		
Power supply voltage	$\pm (15 \pm 0.2)^{\circ}$	V	мо все рт-н z -ч2
Power consumption	300 mW		n Noise
Temperature range of operation	-30°C ÷ +50	°с	spectral
Outer dimensions (without prominent parts)	1 = 400 mm		density
	d = 32 mm		
Length of the output cable	0.7 m		
Weight	≤320 g		
			0.1 1 10 100 1000 10000 T, Hz

EGLE Esperia Geo-magnetometer for Low frequency Experiment



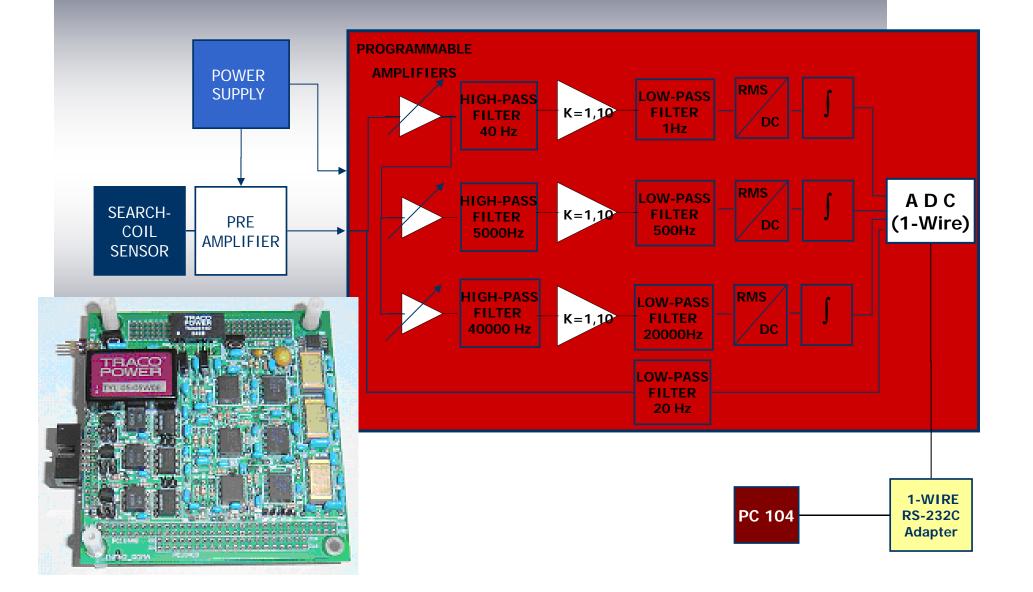
Search-coil frequency band : 0.5 Hz ÷ 50 kHz

Channel	Frequency band	Type of data
Channel A	≤ 40 Hz	Integrated r.m.s
Channel B	(0.5 ÷ 5) kHz	Integrated r.m.s
Channel C	(20 ÷ 40) kHz	Integrated r.m.s
Channel D	≤ 10 Hz	Raw data



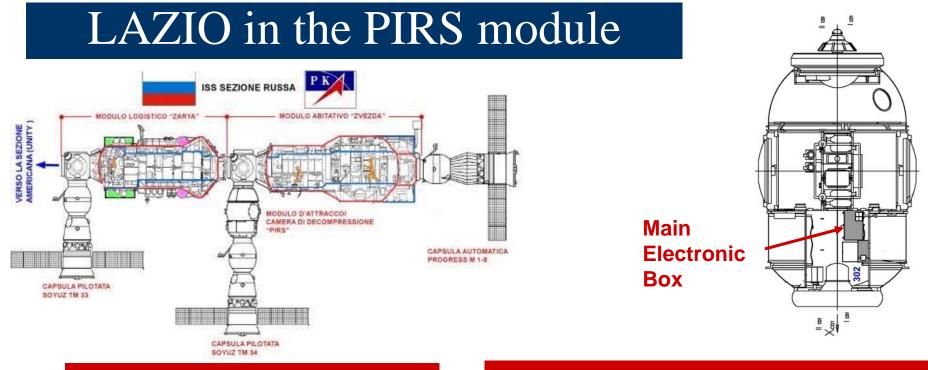
ISS inclination 51.6°, altiyude \approx 350-450 km.

gle Data Acquisition System

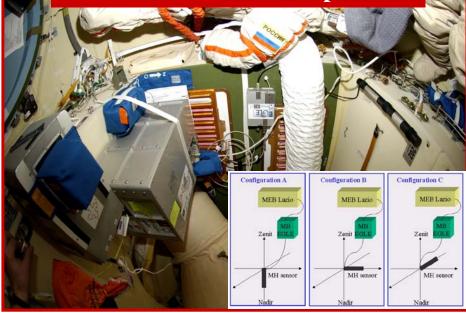








ENEIDE Mission 15-25 April 2005



Increment 14 (Autumn 2006-Spring 2007)







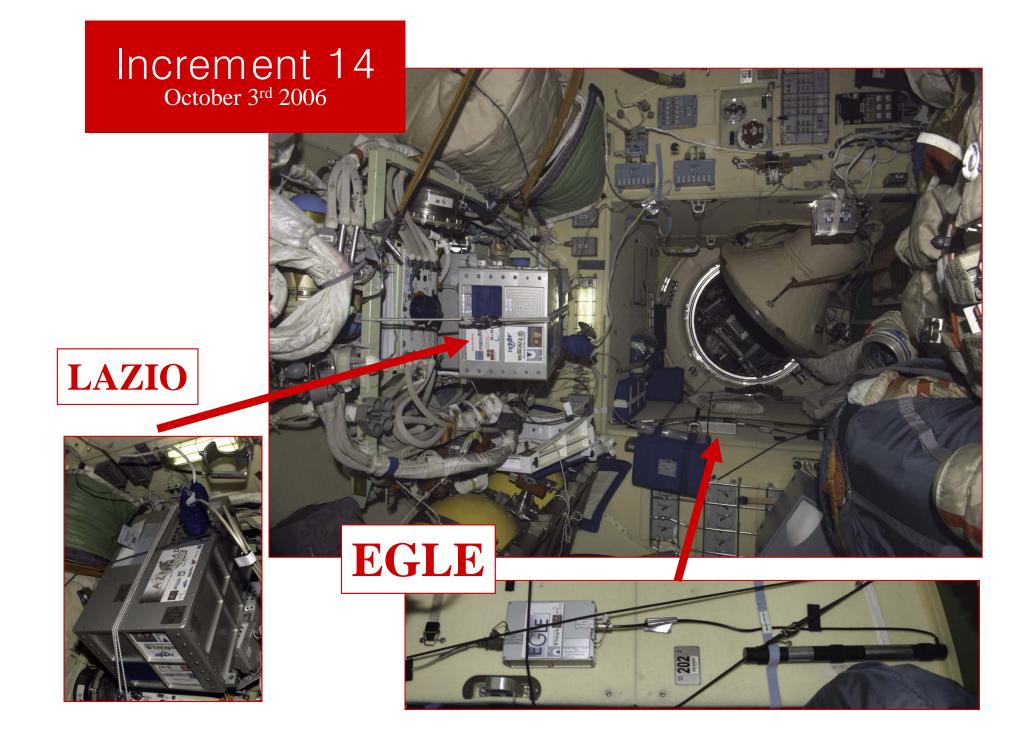
for a Low-frequency wave Experiment



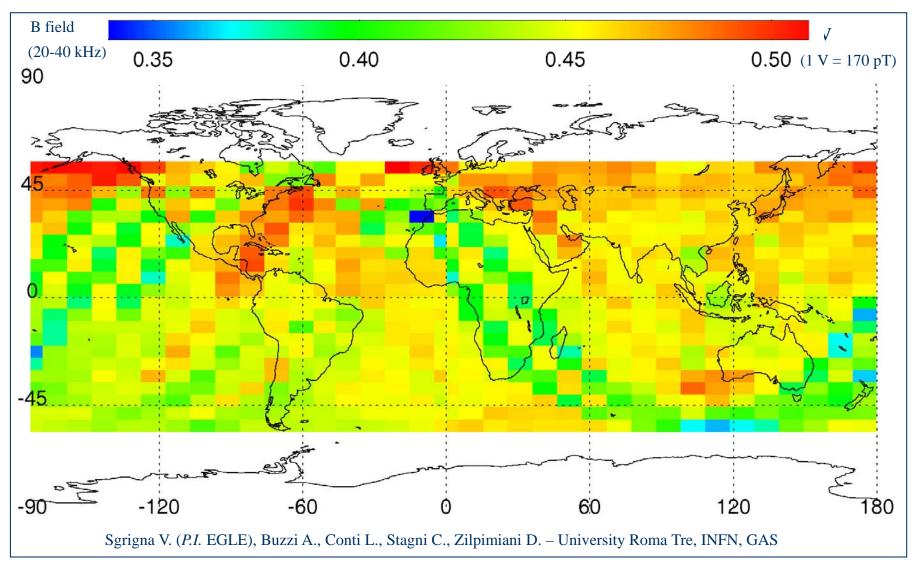
LAZIO

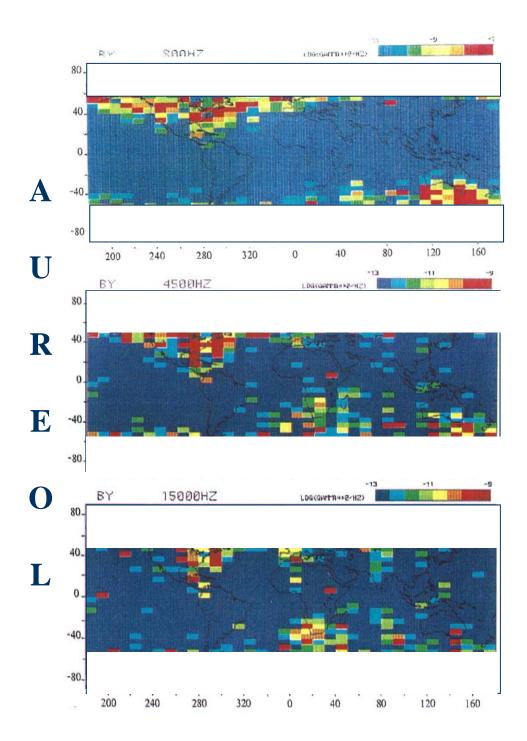
Z X Y = Pirs Axis ~ Nadir)

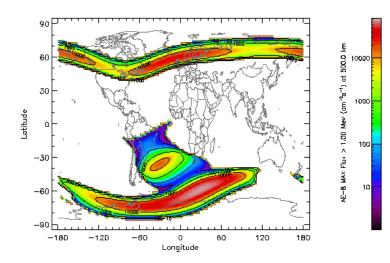


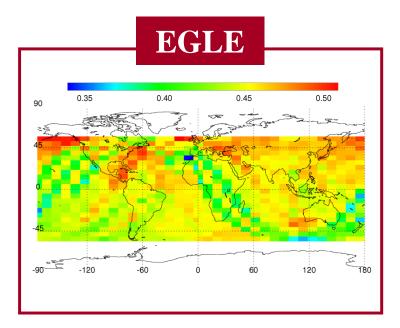


Egle Channel C geographic map

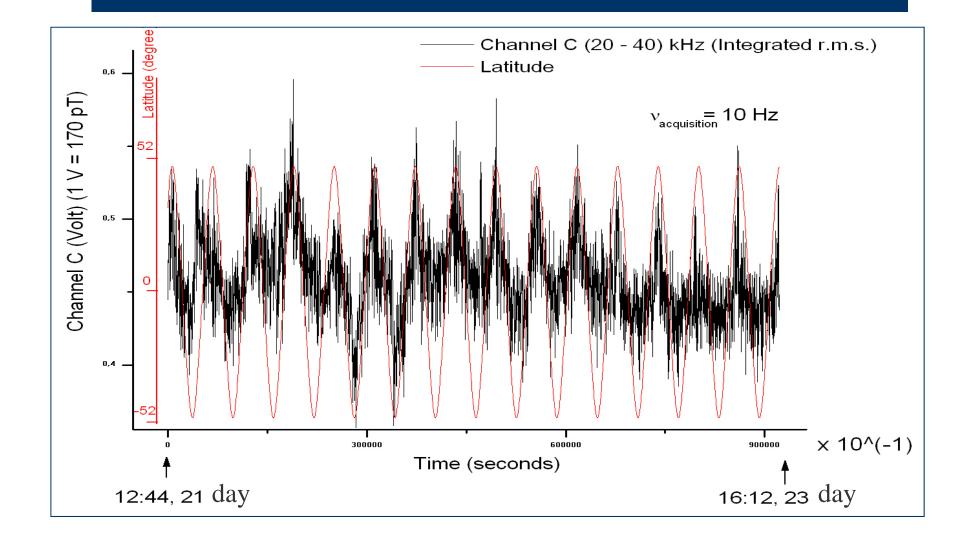








EGLE magnetic field data



Italian participants to the CSES collaboration



- INFN (Sections of Perugia, Rome, Trento, Bologna, Frascati)
- ASI Italian Space Agency
- Perugia University (Perugia)
- Trento University (Trento)
- "Tor Vergata" University (Rome)
- UNINETTUNO University (Rome)
- IFSI / INAF (Rome)
- **CISAS** ("G.Colombo", Padua)











VOLTA Electric Field Detector (EFD) for the CSES satellite

• The EFD is constituted by **4 spherical sensors (probes)** coupled with the amplifying electronics and mounted on **4 deployable booms of 5 meter long each**.

• The continuous measurement of the **electric field vector components** from about DC / ULF up to HF (~15MHz) is obtained from the **measurement of the potential of the four probes which floats at the plasma potential**.

• Therefore, the electric field measurements will be affected by several effects such as: *limitations of the electronics*, *spacecraft electromagnetic noise*, *different work functions of the probes*, and *errors affecting the knowledge of position of each probe*.

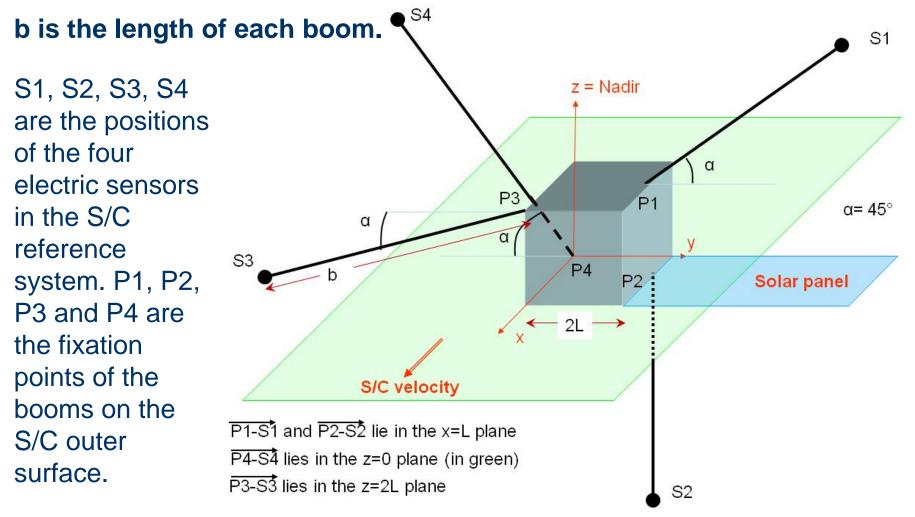
• All the hardware and the measurement procedures, suitable to reduce the electronic noise, and to achieve the required dynamical range & resolution, will be adopted.

• The measurements along the three axes will be carried out in both the operating modes: **survey and burst** modes.

- Particular care will be put in compensating effects due to variations of the S/C <u>temperature</u>.
- Any effort will be done to keep errors (stemming from different work functions of the probes) below levels defined by the required dynamical range & resolution.

VOLTA - EFD booms orientation

The spacecraft (S/C) is schematised as a cube of edge length 2L. The x (parallel to the S/C velocity), z (points to nadir) and y axes, of the S/C reference system, have their origin at one of the cube's vertices.



Electric field resolution & booms stability

Errors affecting the **position of each probe** can stem from two different effects, which may present themselves at the same time:

✓ variations the dipoles length (with respect to their nominal value) due to booms oscillations (around their nominal direction),

&

\checkmark misalignments of the x, y and z axes.

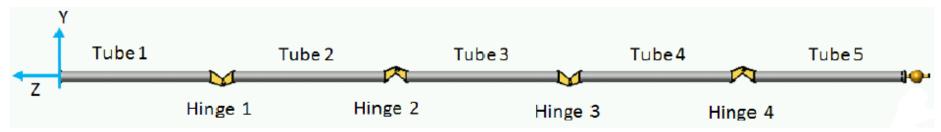
A **constant deviation** of the booms, from their nominal position, would result in a **systematic error** (which would not concern us very much, if we look for variations of the quasi-DC field).

Conversely, **oscillations of the booms**, near their nominal position, have to banalysed with care, in order to understand what effect we should expect:

- one or more lines (at discrete and well-resolved frequencies) or
- noise continuously covering a frequency band.

VOLTA - EFD booms

Figure shows the scheme of a deployed robotic boom.



•The robotic booms ensure high deployment reliability, and high stability of their nominal position with respect to vibrations and misalignments.

•Moreover the robotic booms deployment mechanism preserves the external booms coating and allows to protect cables inside to the boom tubes.

•Consequently the robotic booms constitute one of the **best solutions** to be adopted for the EFD detector on the CSES satellite.

•Nevertheless, to **optimise volume** and **mass budget**, some alternative solutions, such as **collapsible hinge booms** and **Stacer booms**, are under evaluation by the CSES collaboration.

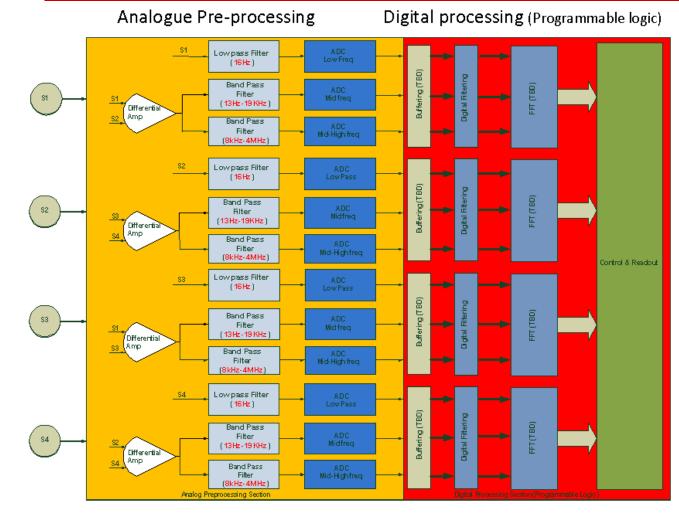
VOLTA - EFD main specifications

Frequency bands	Sampling frequency	Dynamic range	
DC - 10 Hz	100 Hz	$\geq 120 \text{ dB}$	
10 Hz – 2 kHz	$\geq 5 \text{ kHz}$	\geq 90 dB	
2 kHz – 20 kHz	$\geq 50 \text{ kHz}$	\geq 90 dB	
20 kHz – 3.5 MHz	$\geq 10 \text{ MHz}$	\geq 70 dB	
3.5 kHz – 15 MHz	$\geq 40 \text{ MHz}$	\geq 70 dB	

	Total data rate (Gbits/day)	
ICE	~ 6	
VOLTA	~ 42	

Resolution	$1 \mu V/m (DC - 10 Hz)$		
Power supply voltage	+ 28 V		
Operating temperature (inside)	-10 °C ÷ +45 °C		
Operating temperature (outside)	- 30 °C ÷ + 60 °C		
Mass for each probe	$\leq 0.5 \text{ kg}$		
Total Mass budget	$\leq 10 \text{ kg}$		
(including electronic box,			
but excluding booms)			
Power consumption	~ 25 W (TBC)		
Data bus	LVDS		
Operating mode	Survey and Burst		

VOLTA - EFD architecture



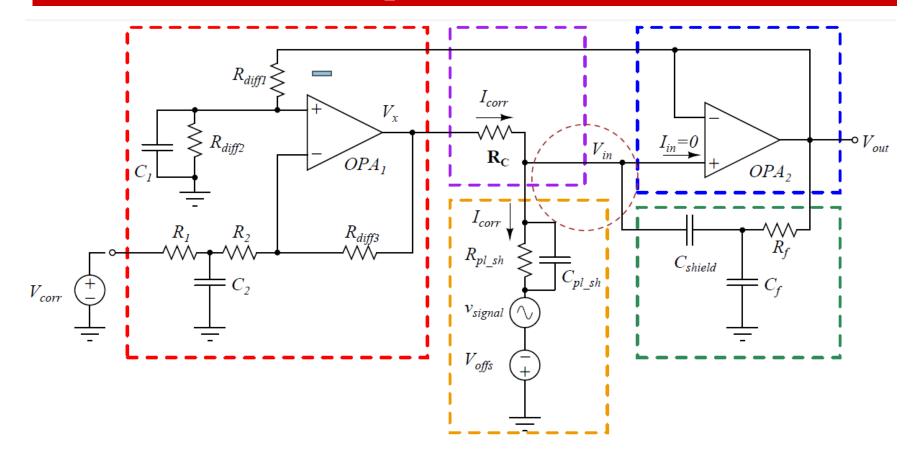
 No analogue multiplexers are adopted for selecting different inputs from sensors (4 DC-16 Hz potential waveforms will be always available) or from differential amplifiers providing different electric field waveforms above 13Hz (DEMETER adopted analogue multiplexer).

• This choice should ensure that **signals are cleaner and have a good S/N ratio**.

• Concerning the **analogue electronics** a simple **cold redundancy** will be adopted: if a section fails, it will be powered off and its redundant section will be powered on.

• Concerning the digital electronics we will adopt a triple redundancy.

VOLTA - EFD probe: functional scheme



Red box (left) is the circuit of the injection of current.
Purple box (top - middle) is the injection current correction.
Orange box (bottom - center) represents the plasma model to be investigated.
Blue Box (top - right) is the voltage-follower circuit with high input impedance.
Green box (lower right) is the filter on the intermediate screen.
The dashed circle in brown represents the sphere outside of the probe.

ICE – DEMETER vs VOLTA – CSES

- ✓ The preamplifier is installed on a small board inside a metallic box (ground shielding) located inside the spherical probe
- ✓ Operational amplifier is a low noise & high input impedance
- ✓ The EDF preamplifier board adopts several improvements with respect to the ICE-DEMETER solution constituted by: a reduction of the noise and tuning of some other parameters.

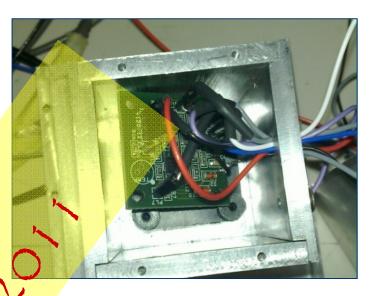
	ICE – DEMETER	EFD – CSES satellite
OPA	OPA131	OPA140
1/f noise	60 <i>n</i> V/Sqrt(Hz) @ 1 <i>Hz</i>	12 nV/Sqrt(Hz) @ 1Hz
band width	4 MHz	11 MHz
Slew-rate	10 V/s	20 V/s

Input resistance of about $10^{13} \Omega$, input capacitance of about pF.

VOLTA- EDF Front-end electronics - Preliminary setup

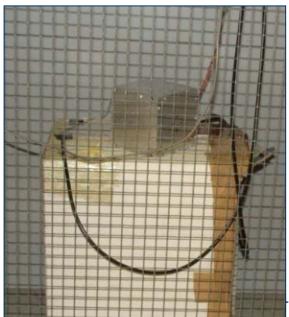


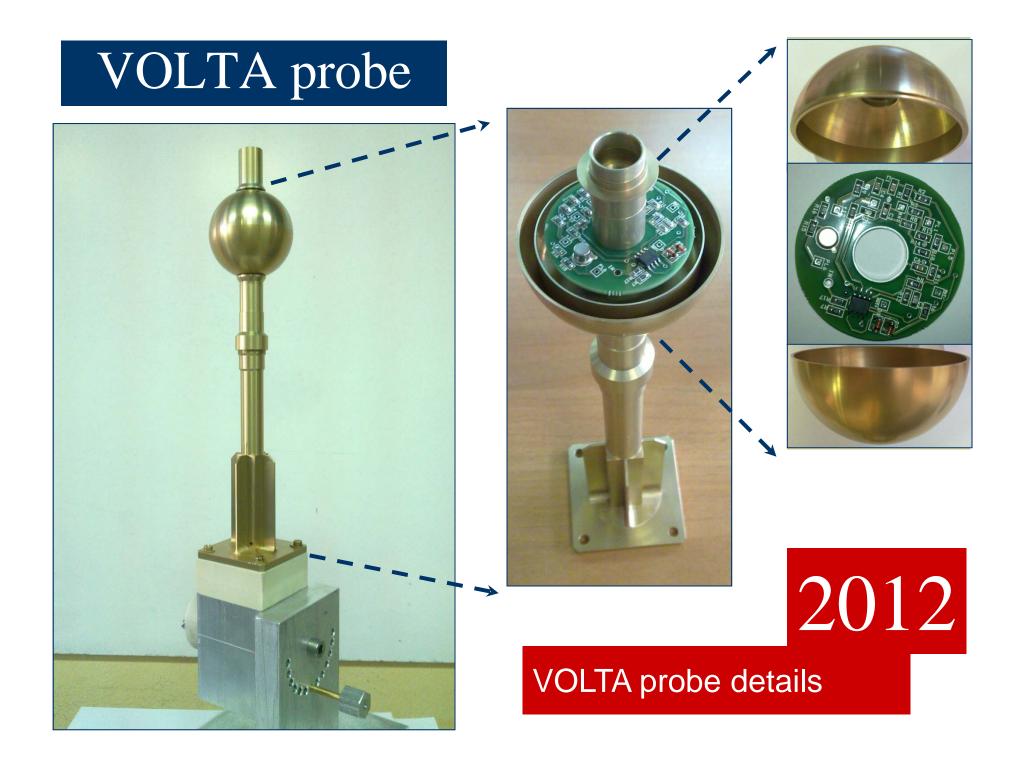
Two metal boxes emulate the probe and its internal shielding. The capacitance between them is 18 pF.

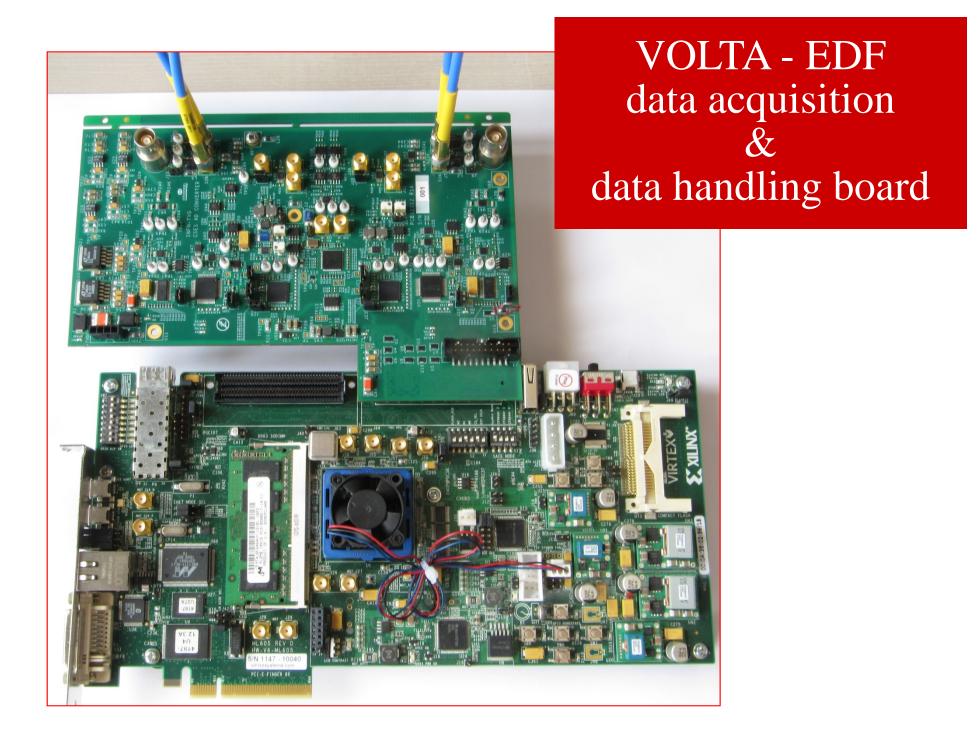


The box which emulates the probe has been placed in a shield connected to ground in order to isolate the probe from external noise. The capacitance between probe and grounded shield is 6 pF.









Simulated Plasma characteristics

• Plasma impedance is a function of the plasma temperature and density conditions along the orbit.

• To take into account the variability of the plasma impedance, 3 different conditions have been considered: **configuration A** and **B** (with **extreme high density and temperature**) and the so called **configuration C** (intermediate condition).

Configuration	n (m ³)	T (K)	$R(\Omega)$	C (pF)
А	1012	3000	67	29.0
В	10 ⁹	1000	970	4.8
С	1010	2000	660	6.6

VOLTA - EDF probe test inside an anechoic chamber



VOLTA - EDF probe tested inside a Faraday chamber



In order to test the front-end electronics under conditions similar to those which will be met in orbit, we consider 3 representative plasma conditions.

Configuration

А

в

С

n (m³)

1012

 10^{9}

1010

T (K)

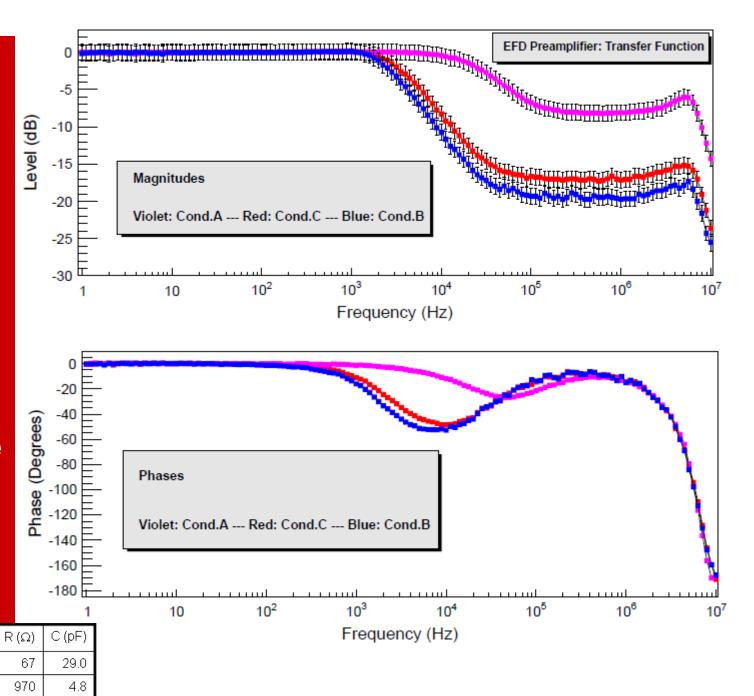
3000

1000

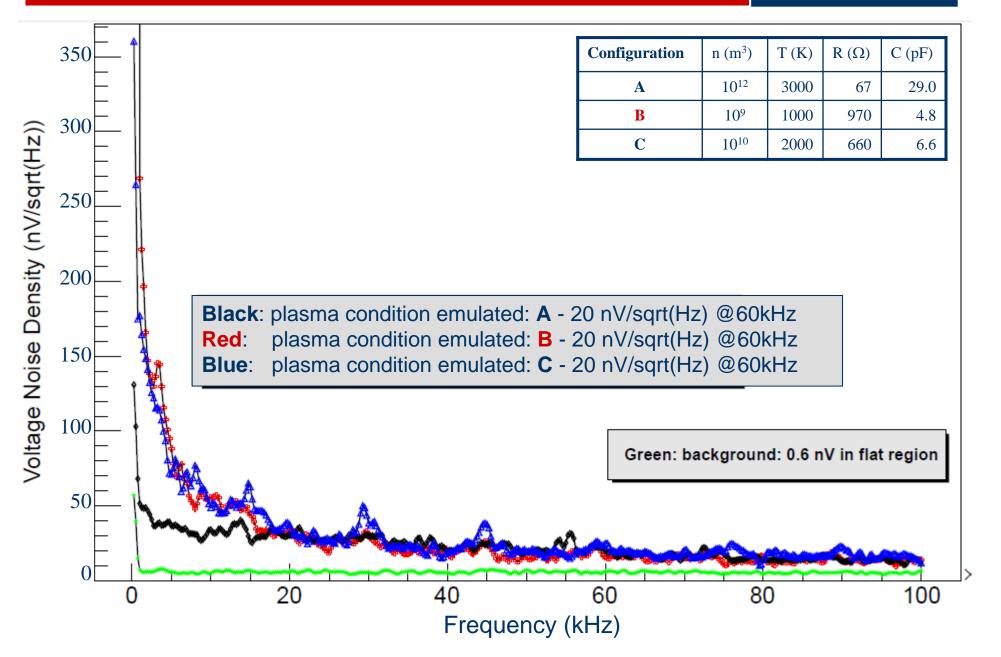
2000

660

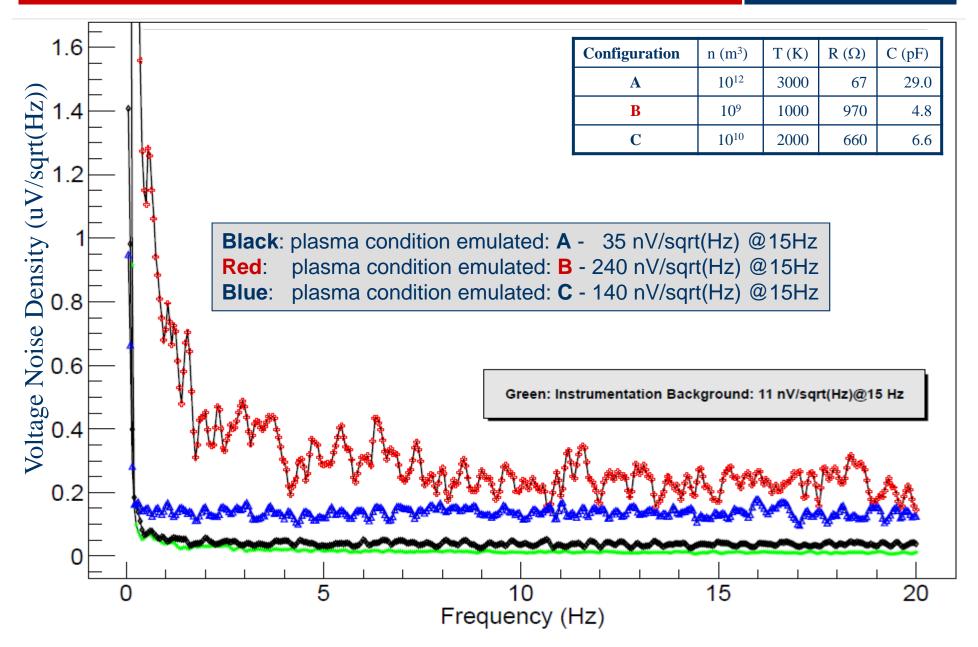
6.6



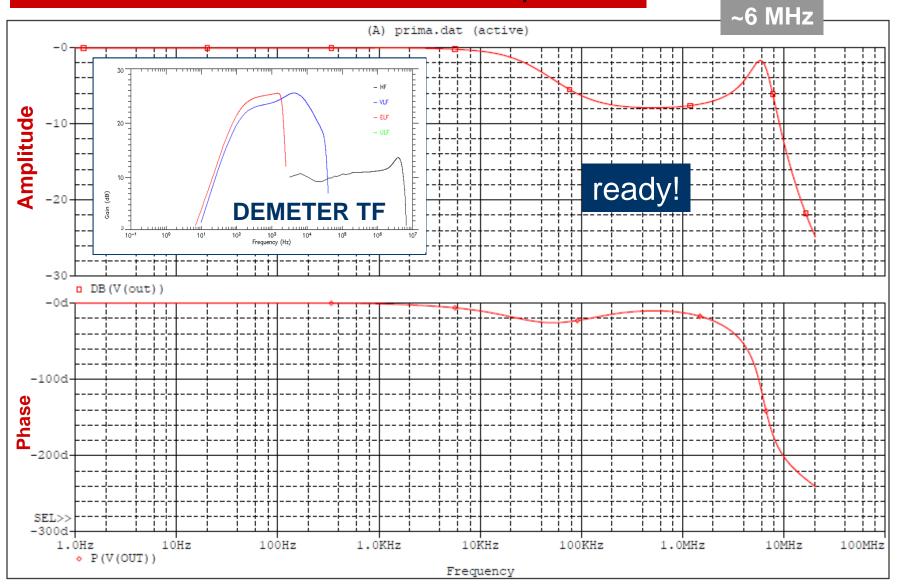
EDF Preamplifier – Voltage Noise Density vs Frequency BW (0-100kHz)



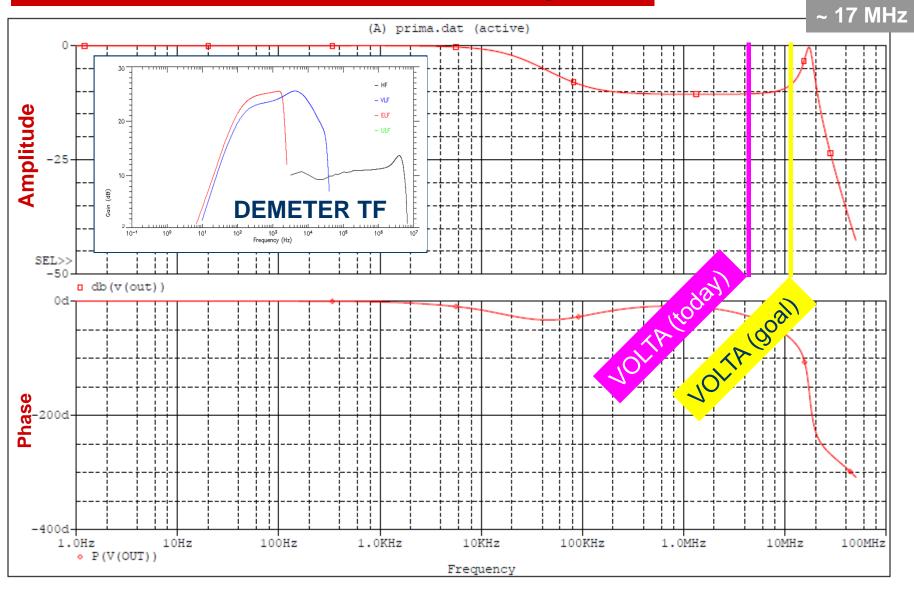
EDF Preamplifier – Voltage Noise Density vs Frequency BW (0.25-20Hz)



VOLTA - EDF Transfer Function: at present



VOLTA - EDF Transfer Function: the goal !!!



OVER- preliminary

- It has been added a differential output to the preamplification system aimed to increase the rejection of the common mode of the induced disturbances.
- At this purpose it has been adopted the OPAMP THS4131 as single-ended-differential converter.
- It whole system is under test to evaluate the intrinsic and induced noise and to verify advantages and disadvantages of such a methodology.



thank you